

IN THE SPECIFICATION

Please replace the paragraph beginning at page 12, line 15, with the following
rewritten paragraph:

In the gas turbine of the sixth embodiment, the curvature from the back side 6 throat SO to the rear edge 6 of the final stage moving blade 4S is defined in a ~~range~~range of 0 to 0.15. As a result, the gauging ratio of the final stage stationary blade 4C of the gas turbine in the second embodiment and the exit angle a2 ratio of the final stage stationary blade 4C of the gas turbine in the third embodiment can be determined appropriately.